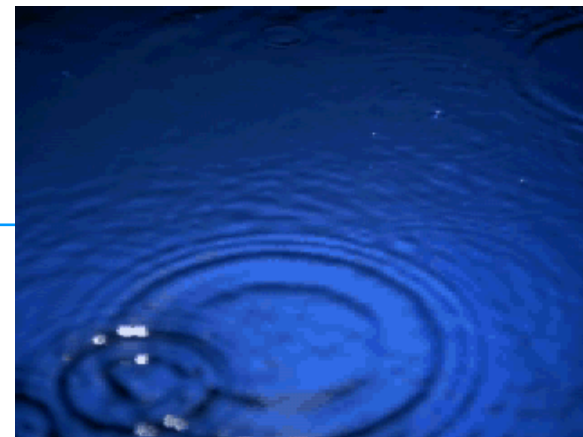


# *Global Precipitation Measurement*

*System Requirements Review*

*Mission Overview*

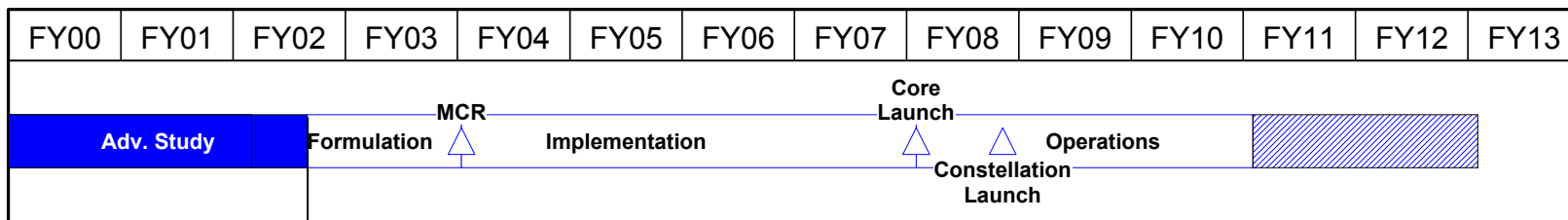


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- *EOS-9 Endorsed at EOS Planning Workshop, 1998*
- *Design Center Studies, 1999-2001*
  - 2 IMDC Sessions
  - 2 ISAL Runs
  - 1 Team X Study
  - 4 RSDO Studies
  - 1 Core In-House Feasibility Study
- *FY01 Advanced Study*
  - Examined Options for All Elements
  - Refined Recommended Approach
- *FY02 Begin Formulation*
  - Partnerships
  - Programmatic and Management Planning
  - System Requirements Definition



You are here



**OBJECTIVE:** Understand the Horizontal and Vertical Structure of Rainfall and Its Microphysical Element. Provide Training for Constellation Radiometers.

**OBJECTIVE:** Provide Enough Sampling to Reduce Uncertainty in Short-term Rainfall Accumulations. Extend Scientific and Societal Applications.

### Primary Satellite

- Dual Frequency Radar
- Multi-frequency Radiometer
- H2-A Launch
- TRMM-like Spacecraft
- Non-Sun Synchronous Orbit
- ~65° Inclination
- ~400 km Altitude
- ~4 km Horizontal Resolution
- 250 m Vertical Resolution

### Precipitation Validation Sites

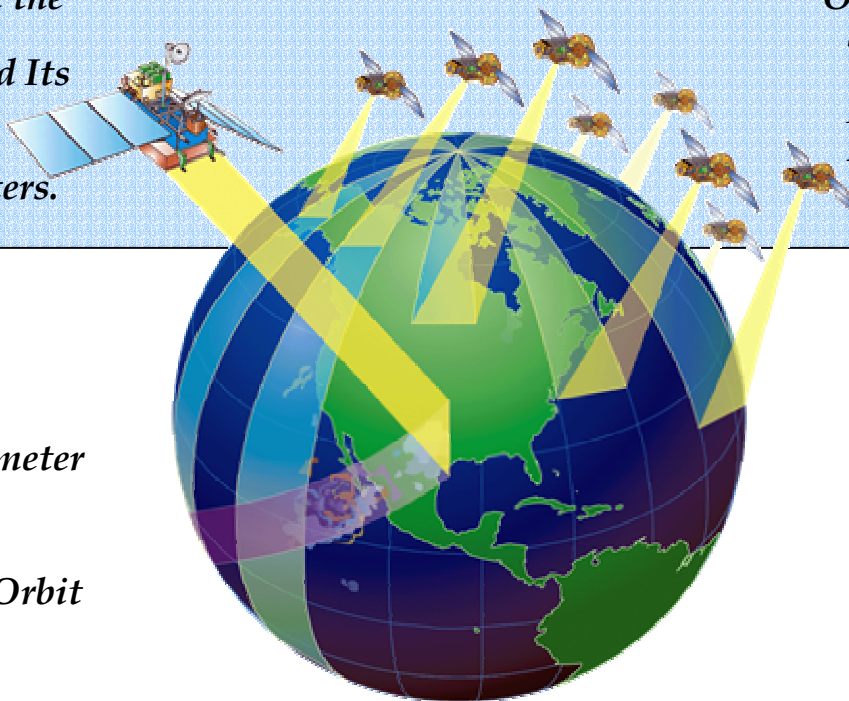
- Ground Truth and Calibration
- Cooperative International Research

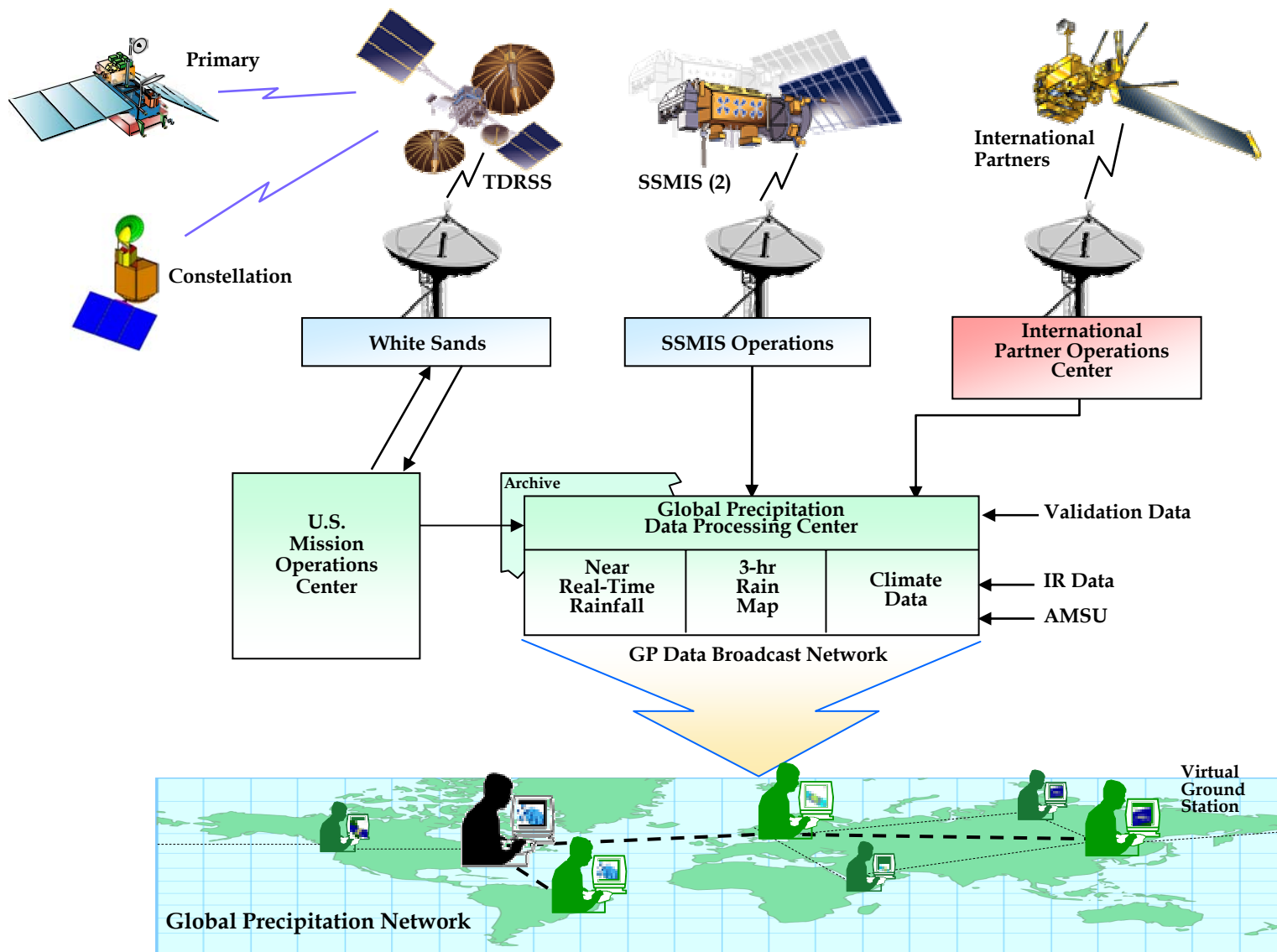
### Constellation Satellites

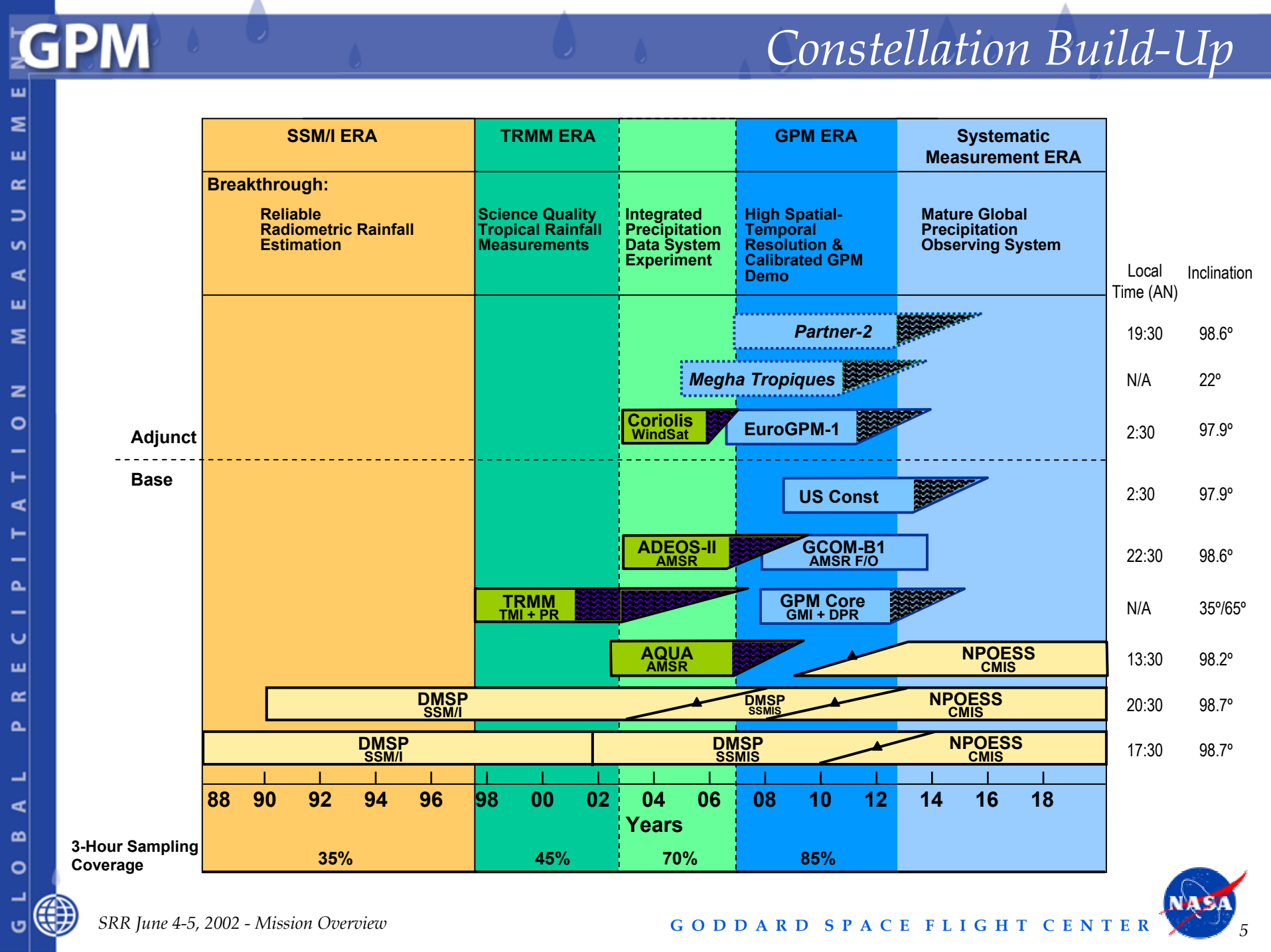
- Multiple Satellites with Microwave Radiometers
- Sampling Sufficient to Resolve the Diurnal Cycle
- Sun-Synchronous Polar Orbits
- ~600 km Altitude

### Global Precipitation Processing System

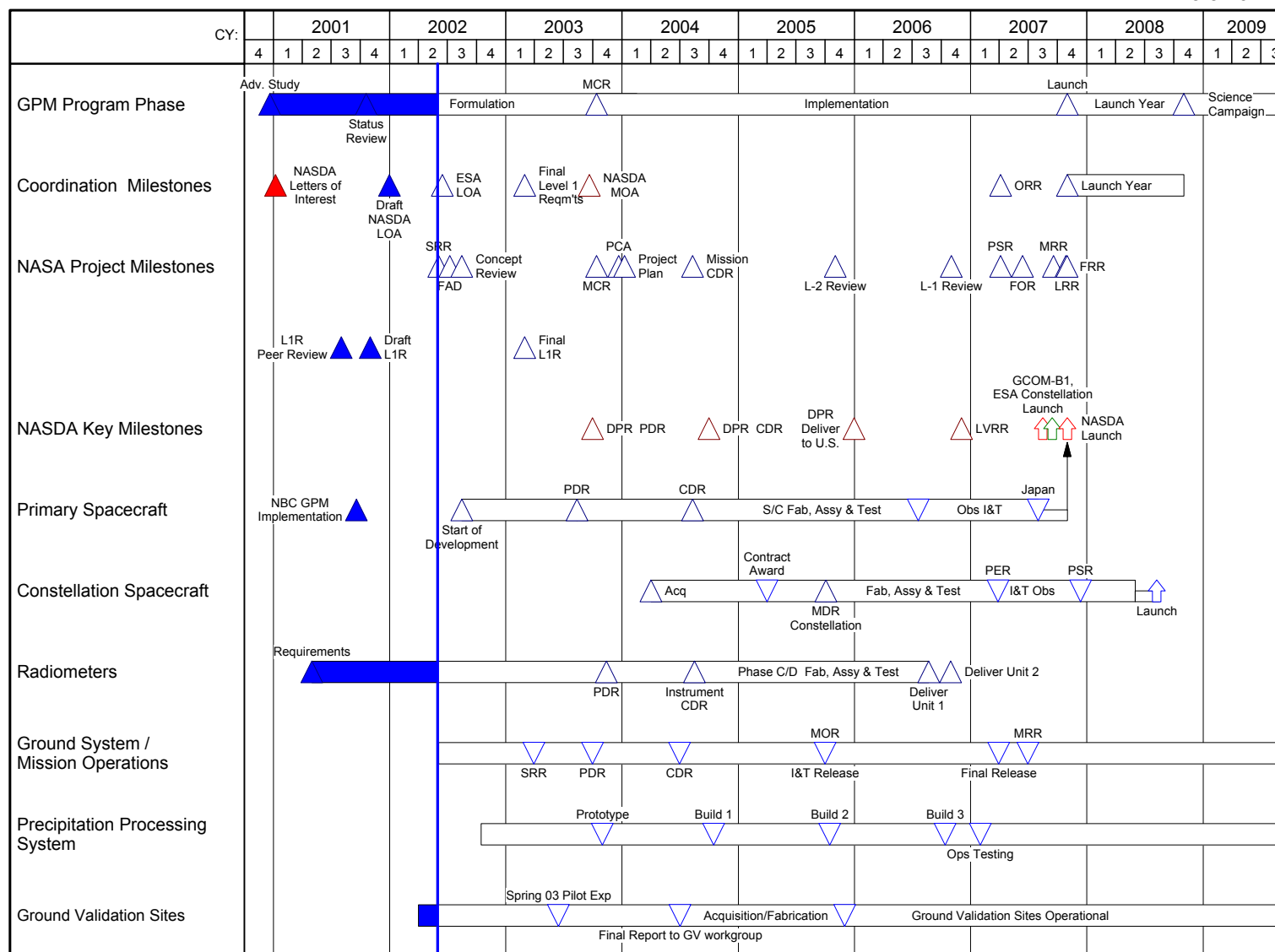
- Capable of Producing Global Precipitation Data Products from Diverse Sensors and Sources
- Cooperative International Partnerships







5/31/02



- ***International Partnerships***

- *Japan critical to mission success*
  - *Follows same model as TRMM*
- *Data streams*
  - *No hardware exchange planned*
  - *Base mission requires only NASA and IPO data streams*

- ***Science Data Processing Multiple Asynchronous Data Streams from Diverse Sensors***

- *Evolve from TSDIS design*
- *SEEDS/IPDS/PPS accelerated implementation retires GPM risk early*
- *Trial runs with*
  - *AMSR*
  - *Megha-Tropiques*
  - *Coriolis (possible)*



- ***Names***
  - *Core = Primary spacecraft*
- ***Core Spacecraft Mass Allocation In-Flux***
  - *NASA currently using 3200 kg, NASDA recent update of H-IIA 3000 kg*
  - *Solutions Exist*
- ***Calibration and Error Characterization (Ground Instrumentation)***
  - *Approach needs better definition before preliminary design starts*
  - *Not significant on space segments*
  - *Science Working Group established*
  - *Field experiments planned*

